

Duality-Based Stochastic Policy Optimization for Estimation with Unknown Noise Covariances

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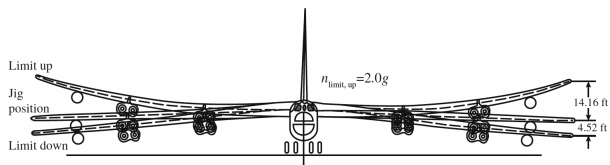
a motivating example

Flexible wing aircraft:

- Boeing 787 Dreamliner



- B-52¹

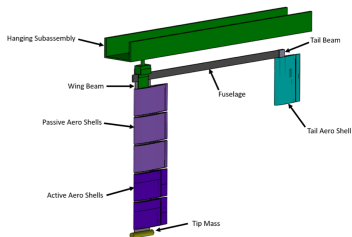


¹Vas and Farokhi, “Introduction to Transonic Aerodynamics”

a motivating example

- (Large) Model for Aeroelastic Response to Gust Excitation

MARGE²



LARGE³



- Goal: **estimate** aeroelastic modal displacements with **approximate** model and **unknown noise covariances**

²[Quenzer, et. al. '19]

³[Nguyen, et. al. '20]

a few estimation approaches

- **full identification:** [Zhang, et. al. '20] IEEE access
 - [Odelson '03], [Rajamani, et. al. '09] Autocovariance LS
 - [Matisko, et. al. '10], [Akesson, et. al. '08], [Dunik, et. al. '09],...
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 - [Lale, et. al. '20], [Tsiamis, et. al. '23] → logarithmic **regret**
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how to learn optimal estimation policy from output data?

estimation-control **Duality** →

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control related works: [Fazel, et. al. '18], [Bu, et. al. '19], [Mohammadi, et. al. '21], [Zhao, et. al. '21], [Tang, et. al. '21], [Talebi, et. al. '22]...

problem formulation

Consider

$$x(t+1) = Ax(t) + \xi(t)$$

$$y(t) = Hx(t) + \omega(t)$$

$$\mathbb{E}\xi(t)\xi(t)^\top = Q, \quad \mathbb{E}\omega(t)\omega(t)^\top = R, \quad \mathbb{E}x(0)x(0)^\top = P_0, \quad \xi(t) \perp \omega(s)$$

(A, H) **known** (possibly **unstable**) and Q, R, P_0 **unknown**

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Problem: given observations $\mathcal{Y}(t) = \{y(0), y(1), \dots, y(t-1)\}$, find

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- optimal MSE estimate \hat{x}_L is Kalman filter:

$$\hat{x}_L(t+1) = A\hat{x}_L(t) + L(t)(y(t) - H\hat{x}_L(t))$$

with Kalman gain $L(t) = AP(t)H^\top(HP(t)H^\top + R)^{-1}$ and

$$P(t+1) = \text{Riccati}(P(t), Q, R), \quad P(0) = P_0$$

estimation-control duality

- minimize an **LQR cost** over the **backward adjoint** dynamics

$$\min_{u(t)} \leftarrow J_T(u_t) = z^\top(0)P_0z(0) + \sum_{t=1}^T z^\top(t)Qz(t) + u^\top(t)Ru(t)$$

$$s.t. \quad z(t) = A^\top z(t+1) - H^\top u(t+1), \quad z(T) = a$$

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- the reparameterized problem: $J(L) := \lim_{T \rightarrow \infty} J_T(L)$

$$\min_L \leftarrow J(L) := \text{tr} [X_{(L)} H^\top H],$$

$$s.t. \quad X_{(L)} = A_L X_{(L)} A_L^\top + Q + L R L^\top$$

$$L \in \mathcal{S} := \{L \in \mathbb{R}^{n \times m} : \rho(A - LH) < 1\}$$

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- define squared estimation error $\text{SE}(L, \mathcal{Y}_T) := \|y(T) - H\hat{x}_L(T)\|^2$

$$J(L) \stackrel{\text{Duality}}{=} \lim_{T \rightarrow \infty} \mathbb{E}[\text{SE}(L, \mathcal{Y}_T)]$$

Stochastic Gradient Descent for estimation

the algorithm:

[Feed observations to the built stochastic oracle] $\mathcal{Y}_T \mapsto \nabla_L \text{SE}(L, \mathcal{Y}_T)$

[Approximate a biased noisy gradient] $\nabla \hat{J}_T(L) = \frac{1}{M} \sum_{i=1}^M \nabla_L \text{SE}(L, \mathcal{Y}_T^i)$

[Run M -batch SGD] $L_{k+1} \leftarrow L_k - \eta_k \nabla \hat{J}_T(L)$

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• comparison: control vs optimal estimation

Problem	Parameters			Constraints	Gradient Oracle	
	cost value	Q and R	A and H	stability \mathcal{S}	model	biased
⁴	known	known	unknown	yes	$\mathbb{E}J(L + r\Delta)\Delta$ $\Delta \sim U(\mathbb{S}^{mn})$	yes
Estimation (this work)	unknown	unknown	known	yes	$\mathbb{E}\nabla \text{SE}(L, \mathcal{Y})$ $\mathcal{Y} \sim \text{output data}$	yes
Vanila SGD	*	*	*	no	$\mathbb{E}\nabla \text{SE}(L, \mathcal{Y})$ $\mathcal{Y} \sim \text{data dist.}$	no

⁴[Fazel; et. al. '18] and similar LQR control works.

Convergence [Informal]

Consider observable (A, H) , bounded noise $\xi(t), \omega(t)$, and small stepsize. Then (w.p. $\geq 1 - \delta$), SGD converges **linearly** and **globally** to ϵ -optimal Kalman gain if

- # of trajectories $\geq O(\ln(1/\delta)/\epsilon^2)$
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—SGD convergence with:

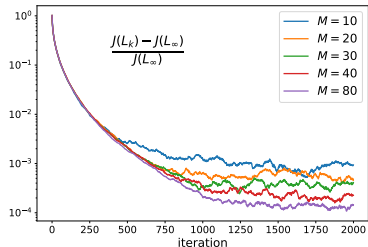
biased gradient, **locally Lipschitz** and **stability constraint**

—statistical bounds: with high prob

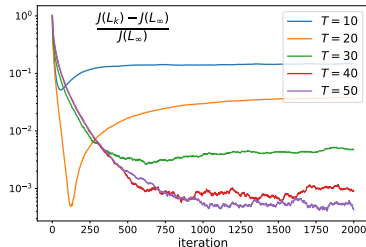
$$\|\nabla \hat{J}_T(L) - \nabla J(L)\| \leq \underbrace{\|\nabla \hat{J}_T(L) - \nabla J_T(L)\|}_{\text{concentration}} + \underbrace{\|\nabla J_T(L) - \nabla J(L)\|}_{\text{truncation} \rightarrow \text{bias}}$$

numerical simulation

- The **average optimality gap** (average over 40 simulations) for different (a) batch-size M and (b) trajectory length T .



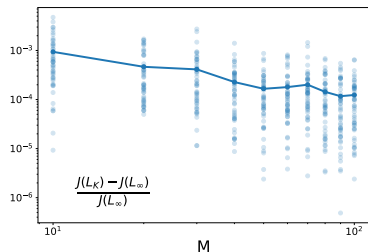
(a)



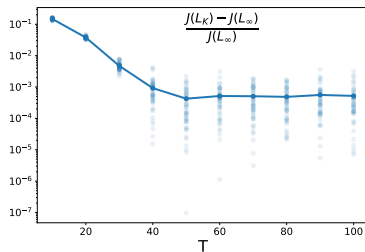
(b)

numerical simulation

- The **optimality gap at final iteration** of every simulation as a function of (a) batch-size M and (b) trajectory length T .



(a)



(b)

- Future directions: “optimal” data usage, and perturbed systems parameters